



Introduction to DAVE-ML

Flight Dynamic Model Exchange using XML

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What is XML?

- XML (like HTML) is based on SGML, an early ISO standard for electronic information storage & exchange
- XML is a flexible, extensible meta-markup language where information is delineated by contextual tags
- XML is not a programming language



XML syntax

- Elements (info surrounded by tag pair)
`<mytag>information</mytag>`
- Attributes (info embedded in tag)
`<mytag important="yes">information</mytag>`
- Empty elements (single tag as flag)
`<useEnglishUnits/>`
- Comments

`<!-- a comment goes here -->`



Required header

- Each XML document should start with a single-line xml declaration:

```
<?xml version="1.0" encoding="US-ASCII" standalone="yes"?>
```

- This identifies it as XML without a external grammar specification



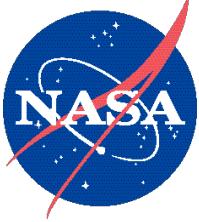
Validation

- XML documents must be *well-formed* (meaning all start- and end-tags are paired and correctly nested, etc.)
- If using an external grammar, an XML document is *valid* if it respects the grammar rules in the external document
- Special XML editors can determine validation (emacs, BBEdit, others)



Document Type Definition (DTD)

- A DTD is an XML document that defines a particular grammar - that is, the allowed elements and attributes and their relationship to each other
- Math-ML, for example, has a DTD that defines allowable mathematical elements
- DAVE-ML consists of an evolving DTD



Math-ML

- Math-ML is a W3C-adopted XML-based grammar for mathematical information
- Includes specifications for both notation (how the equation appears) and content (how the relationship works)
- Should allow useful, interactive, hi-quality mathematical expressions on Web pages
- Supported by Wolfram Research (Mathematica)



AIAA standards

- DAVE-ML proposed to implement draft AIAA standard practice regarding
 - Variable names
 - Function table information
- Additional standards may be rolled in
- DAVE-ML may be offered to AIAA as standard



DAVE-ML grammar

- Intended to encode complete high-fidelity flight model in non-proprietary, facility-independent way
- First step: static models (tables & equations)
 - Aerodynamic model
 - Inertial subsystem models
- Next step:
 - Checkcase data (static & dynamic)
 - Time-history data
- Eventually include dynamic elements



DAVE-ML top-level elements

- **fileHeader** - Contains author, creation date, description, references, modification record
- **variableDef(s)**
 - Describes inputs, outputs, intermediate signals
 - Includes build-up equations with basic math functions
 - Equations can reference table outputs
- **breakpointDef(s)** - Used to store breakpoint sets which may be reused
- **function(s)** - Marries breakpoint sets with independent variables to form output (dependent) variable



fileHeader element details

- author
- fileCreationDate
- description (optional)
- reference(s) (optional)
- modificationRecord(s) (optional)



fileHeader example

```
<?xml version="1.0" standalone="no"?>
<!DOCTYPE DAVEfunc SYSTEM "DAVEfunc.dtd">
<DAVEfunc>
  <fileHeader>

    <author name="Bruce Jackson" org="NASA Langley Research Center" xmlns="@bjax"/>
    <fileCreationDate date="28-MAR-2002"/>
    <description>
      F-16 Aero Data file. Based on Morelli's adaptation of
      Stevens and Lewis' F-16 example [1] described in Garza &amp;
      Morelli's TM [2]. Obtained from E. A. Morelli in the form of
      Matlab scripts [3] &amp; [4]
    </description>

    <!-- ===== -->
    <!-- References -->
    <!-- ===== -->

    <reference refID="REF01" author="Stevens, Brian L. and Lewis, Frank L."
      title="Aircraft Control and Simulation"
      accession="ISBN 0-471-61397-5" date="1992"/>

    <reference refID="REF02" author="Garza, F. R.; and Morelli, E. A."
      title="A Collection of Nonlinear Aircraft Simulations
      in MATLAB" accession="NASA TM-2002-xxxxxx" date="JUN-2002"/>

    <reference refID="REF03" author="Morelli, Eugene A."
      title="f16_aero.m" date="17-JUN-1995"/>

    <reference refID="REF04" author="Morelli, Eugene A."
      title="f16_aero_setup.m" date="17-JUN-1995"/>

  </fileHeader>
```



variableDef element details

- Attributes used to define name, assign unique variable ID, specify units
- Optional attributes include axis system, sign, alias, and symbol (UNICODE)
- Sub-elements
 - `description` (optional)
 - `calculation` (optional - uses mathML 2.0)
 - Empty `<isOutput/>` element signifies intermediate output quantity



variableDef examples (constants)

```
<variableDef name="rtd" varID="rtd" units="deg/rad">
  <description>
    Conversion constant from radians to degrees
  </description>
  <calculation>
    <math>
      <apply>
        <quotient/>
        <cn>180.</cn>
        <cn>3.14159265</cn>
      </apply>
    </math>
  </calculation>
</variableDef>

<variableDef name="xcgr" varID="xcgr" units="fract" initialValue="0.35">
  <description>
    Default location of center of gravity relative to wing leading
    edge, expressed as a fraction of aerodynamic chord length.
  </description>
</variableDef>
```



variableDef examples (inputs)

```
<variableDef name ="True_Airspeed_f_p_s" varID="vt" units= "ft/sec">
  <description>
    True airspeed, ft/sec
  </description>
</variableDef>

<variableDef name="Angle_of_Attack_deg" varID="alpha" units="deg" symbol="#x3B1">
  <description>
    Instantaneous true angle-of-attack, in degrees
  </description>
</variableDef>

<variableDef name=" Angle_of_Sideslip_deg " varID="beta" units="deg" symbol="#x3B2"
  sign="+wind in right ear">
  <description>
    Instantaneous true angle-of-sideslip, in degrees
  </description>
</variableDef>
```



variableDef examples (intermediate calculations)

```
<variableDef name="drdr" varID="drdr" units="">
  <description>
    Normalized rudder deflection.
  </description>
  <calculation>
    <math>
      <apply>
        <quotient/>
        <ci>rdr</ci>
        <cn>30.0</cn>
      </apply>
    </math>
  </calculation>
</variableDef>
```



variableDef example (output calculation)

```
<variableDef name="Cl0" varID="clt" units="" sign="+right wing down">
  <description>
    Basic coefficient of moment around the X-body direction (roll) (+RWD)
  </description>
  <calculation>
    <math>
      <apply>
        <piecewise>
          <piece>
            <apply><minus/><ci>absCl0</ci></apply>
            <apply><lt/><ci>beta</ci><cn>0</cn></apply>
          </piece>
          <otherwise>
            <ci>absCl0</ci>
          </otherwise>
        </piecewise>
      </apply>
    </math>
  </calculation>
</variableDef>
```



breakpointDef element details

- Attributes define unique **bpID**, optional **name** and **units**
- Sub-elements
 - **description** (optional)
 - **bpVals**



breakpointDef examples

```
<breakpointDef name="beta" bpID="BETA1" units="deg">
  <description>
    Angle-of-sideslip breakpoints for basic aero tables
  </description>
  <bpVals>
    0.0, 5.0, 10.0, 15.0, 20.0, 25.0, 30.0
  </bpVals>
</breakpointDef>

<breakpointDef name="beta" bpID="BETA2" units="deg">
  <description>
    Angle-of-sideslip breakpoints for control power tables
  </description>
  <bpVals>
    -30.0, -20.0, -10.0, 0.0, 10.0, 20.0, 30.0
  </bpVals>
</breakpointDef>
```



function elements

- Sole attribute: `name`
- Sub-elements
 - `description` (optional)
 - `provenance` (optional)
 - Two styles of functions:
 - Simple 1-D (`independentVarPts`, `dependentVarPts`)
 - n-D `functionDefn` with `varRefs`



n-D functions

- Three elements:
 - `independentVarRef(s)` using `varID`
 - Optional `min`, `max`, `extrapolate` attributes
 - Listed order is important
 - `dependentVarRef` using `varID`
 - `functionDefn` (with optional name attribute)
 - Two styles of tables
 - Gridded table (orthogonal)
 - Ungridded table (point-by-point)



griddedTable element

- Sole attribute: **name** (optional)
- Sub-elements
 - **breakpointRefs** - list of **bpRef**(s)
 - Optional **confidenceBound** value
 - Multiplicative 3-sigma ± variation
 - **dataTable**
 - Comma-separated list of gridded table values in which the last **bpRef** changes most rapidly



functionDefn example

```
<function name="Basic Cl">
  <description>
    Basic coefficient of rolling moment as a function of angle of attack and sideslip angle
  </description>
  <provenance>
    <author name="Bruce Jackson" org="NASA Langley Research Center" xmlns="@bjax"/>
    <functionCreationDate date="28-MAR-2002"/>
    <documentRef docID="REF01"/>
    <documentRef docID="REF02"/>
    <documentRef docID="REF03"/>
  </provenance>
  <independentVarRef varID="absbeta" min="0.0" max="30.0" extrapolate="neither"/> <!-- Beta breakpoints -->
  <independentVarRef varID="alpha" min="-10.0" max="45.0" extrapolate="neither"/> <!-- Alpha breakpoints -->
  <dependentVarRef varID="absCl0"/>

  <functionDefn name="Cl0_fn">
    <griddedTable name="Cl0_table">
      <breakpointRefs>
        <bpRef bpID="BETA1"/>
        <bpRef bpID="ALPHA1"/>
      </breakpointRefs>
      <dataTable> <!-- Note: last breakpoint changes most rapidly -->
        0., 0., 0., 0., 0., 0., 0., 0., 0.,
        -.001,-.004,-.008,-.012,-.016,-.022, -.022,-.021,-.015,-.008,-.013,-.015,
        -.003,-.009,-.017,-.024,-.030,-.041, -.045,-.040,-.016,-.002,-.010,-.019,
        -.001,-.010,-.020,-.030,-.039,-.054, -.057,-.054,-.023,-.006,-.014,-.027,
        .000,-.010,-.022,-.034,-.047,-.060, -.069,-.067,-.033,-.036,-.035,-.035,
        .007,-.010,-.023,-.034,-.049,-.063, -.081,-.079,-.060,-.058,-.062,-.059,
        .009,-.011,-.023,-.037,-.050,-.068, -.089,-.088,-.091,-.076,-.077,-.076
      </dataTable>
    </griddedTable>
  </functionDefn>
</function>
```



ungriddedTable element

- Sole attribute: `name` (optional)
- Sub-elements
 - Optional `confidenceBound` value
 - Multiplicative 3-sigma ± variation
 - `dataPoint(s)`, each contains a comma-separated list:
 - Independent variable value (in order)
 - Dependent value
 - -3-sigma value (optional)
 - +3-sigma value (optional)
 - refID (optional)



What's next?

- Demo of exchange of aero models
- Feedback to refine grammar
- Develop tools - editors, report generator
- Establish informal working group for refinement
- Work towards AIAA standard
- Establish test cases for new facilities